

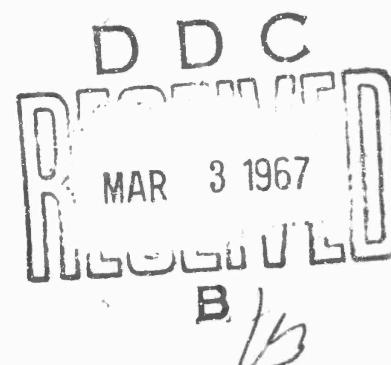
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Report



ON THE EIGENVALUES OF COUETTE  
FLOW IN A FULLY-FILLED  
CYLINDRICAL CONTAINER

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BENJAMIN FRANKLIN PARKWAY • PHILADELPHIA, PENNA. 19103

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Report

ON THE EIGENVALUES OF COUETTE  
FLOW IN A FULLY-FILLED  
CYLINDRICAL CONTAINER

*by*

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## SUMMARY

For a stationary flow in a cylindrical container of the Couette type in an outer radial zone, and of zero velocity in an inner radial zone, the normal mode equations are derived. For negative wave numbers in the  $\theta$ -direction, these equations are found to have a singularity.

The eigenvalues are calculated by initial value methods employing the Runge-Kutta-Gill integration procedure. Values of the dependent function and their derivatives at the singularity are calculated by linear extrapolation coupled with continuity requirements.

Tables of eigenvalues for various slenderness ratios of the cylinder and various radial nodes are given for  $\theta$ -wave numbers of -1.

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## INTRODUCTION

The object of the following investigation is to determine the eigenfrequencies of a viscous fluid, contained in a rotating cylinder, during the initial spin-up period. In order to make the problem tractable, an inviscid fluid is assumed and the fluid is divided into two zones; an outer radial zone in which a Couette type flow prevails and an interior zone which is static. The initial flow distribution, where the interface between the two zones is assumed to decrease from the outer radius to zero in a quasi-static manner, is assumed to be time independent.

## FORMULATION OF THE EIGENVALUE PROBLEM

In the following we are interested in the characteristic frequencies of an inviscid liquid contained in a cylindrical container. The liquid is assumed to have an initial stable motion of the Couette type for an outer radial zone whereas the interior is assumed to be at zero velocity. Thus, let the container be of radius  $a$  and height  $2c$ . Then the stationary flow in the outer zone is given by

$$\begin{aligned} u_o &= 0 & a \geq r \geq b \\ v_o &= aw[(r/a) - (e^2 a/r)]/(1 - e^2) & 0 \leq [e = b/a] \leq 1 \\ w_o &= 0 \end{aligned} \tag{1}$$

In the inner zone,

$$\begin{aligned} u_i &= 0 \\ v_i &= 0 & b \geq r \geq 0 \\ w_i &= 0 \end{aligned} \tag{2}$$

where  $b$  is the radius of the interface between the static interior and the moving exterior;  $u$ ,  $v$ , and  $w$  are the radial, longitudinal and axial components of velocity in a cylindrical coordinate system with the

$z$ -axis aligned along the axis of the cylinder. The origin of the coordinate system is located at the bottom of the cylinder so that the  $z$ -coordinates of the end faces are given by 0 and  $2c$ . The velocity components in the inner and outer zones are distinguished by the subscripts i and o respectively.

For convenience, we consider the two zones, namely, the exterior zone of Couette flow, and the interior zone of zero velocity, separately. These are then coupled by boundary conditions imposed at the interface, where the radial velocities and pressures corresponding to the two regions are required to be equal. Considering the exterior zone first, the continuity and momentum equations with the appropriate boundary conditions are:

$$\begin{aligned} \frac{\partial u_o}{\partial t} + (Q_o \cdot \nabla) u_o - v_o^2/r &= - \frac{\partial}{\partial r} \left( \frac{p_o}{\rho} \right) \\ \frac{\partial v_o}{\partial t} + (Q_o \cdot \nabla) v_o + u_o v_o/r &= - \frac{1}{r} \frac{\partial}{\partial \theta} \left( \frac{p_o}{\rho} \right) \\ \frac{\partial w_o}{\partial t} + (Q_o \cdot \nabla) w_o &= - \frac{\partial}{\partial z} \left( \frac{p_o}{\rho} \right) \\ Q_o \cdot \nabla &\equiv u_o \frac{\partial}{\partial r} + \frac{v_o}{r} \frac{\partial}{\partial \theta} + w_o \frac{\partial}{\partial z} \\ \frac{u_o}{r} + \frac{\partial u_o}{\partial r} + \frac{1}{r} \frac{\partial v_o}{\partial \theta} + \frac{\partial w_o}{\partial t} &= 0 \end{aligned} \quad (3)$$

The boundary conditions are

$$r = a; u_o = 0$$

$$z = 0, 2c; w_o = 0$$

Moreover, at the interface, we have

$$u_{of} = u_{if}; p_{of} = p_{if} \quad (4)$$

where subscript f refers to the interface.

Substituting the following,

$$\begin{aligned} u_0^* &= u_0/aw; v_0^* = v_0/aw; w_0^* = w_0/aw \\ r^* &= r/at; z^* = z/at; t^* = wt; \theta^* = \theta \\ p^*/\rho^* &= p/\rho a^2 \omega^2 \end{aligned} \quad (5)$$

and dropping the asterisk, the stationary flow is given by

$$\begin{aligned} u_0 &= 0 \\ v_0 &= [r - e^2/r]/[1 - e^2] & 1 \geq r \geq e \\ w_0 &= 0 \end{aligned}$$

The boundary conditions now become

$$\begin{aligned} r = 1; u_0 &= 0 \\ z = 0, 2c/a; w_0 &= 0 \end{aligned} \quad (6)$$

At the interface, they are

$$u_{of} = u_{if}; p_{of} = p_{if} \quad (7)$$

Here  $u_{if}$  and  $p_{if}$  are the appropriate non-dimensional radial velocity and pressure prevailing in the flow field of the inner zone at the interface.

From the non-dimensional form of Equation 3,

$$\begin{aligned} \frac{\partial p_o}{\partial r} &= \frac{v^2}{r} \\ \frac{\partial p_o}{\partial \theta} &= 0 \\ \frac{\partial p_o}{\partial z} &= 0 \end{aligned} \quad (8)$$

where

$$\begin{aligned} p_o &= p_o/\rho \\ v &= [r - e^2/r]/(1 - e^2) \end{aligned} \quad (9)$$

Integrating,

$$p_o = \int_e^r (v^2/y) dy + p_1 \quad (10)$$

where  $P_1$  is the equilibrium pressure at the interface.

We now assume perturbations due to a radial displacement,  $\eta$ , of the interface from its equilibrium position corresponding to  $r = e$ .

$$\begin{aligned} u_o &= u'_o \\ v_o &= V + v'_o \\ w_o &= w'_o \\ P_c &= \int_e^r \frac{V^2(y)}{y} dy + P'_o + P_1 \end{aligned} \quad (11)$$

Substituting the perturbations into the dimensionless Euler's equations of motion, and neglecting second and higher order quantities,

$$\begin{aligned} \frac{\partial u'_o}{\partial t} + \frac{V}{r} \frac{\partial u'_o}{\partial \theta} - \frac{2V}{r} v'_o &= - \frac{\partial P'_o}{\partial r} \\ \frac{\partial v'_o}{\partial t} + u'_o \frac{dV}{dr} + \frac{V}{r} \frac{\partial v'_o}{\partial \theta} + \frac{u'_o V}{r} &= - \frac{1}{r} \frac{\partial P'_o}{\partial \theta} \\ \frac{\partial w'_o}{\partial t} + \frac{V}{r} \frac{\partial w'_o}{\partial \theta} &= - \frac{\partial P'_o}{\partial z} \\ \frac{u'_o}{r} + \frac{\partial u'_o}{\partial r} + \frac{1}{r} \frac{\partial v'_o}{\partial \theta} + \frac{\partial w'_o}{\partial z} &= 0 \end{aligned} \quad (12)$$

The boundary conditions become

$$\begin{aligned} r = 1; u'_o &= 0 \\ z = 0, 2c/a; w'_o &= 0 \end{aligned} \quad (13)$$

The boundary conditions due to displacement of the interface are obtained in the following manner:

Let the interface be at

$$r = e + \eta \quad (14)$$

Substituting into the last of Equations 11 and neglecting second and higher order terms,

$$P_{of} = P_1 + P'_{of} \quad (15)$$

At the interface, we have from 14

$$u'_o = \frac{\partial \eta}{\partial t} \quad (16)$$

where second and higher order quantities in the perturbations have been ignored.

We now consider the interior zone. The Euler's equations of motion and the equation of continuity are once again given by Equation 3 in a non-dimensional form. The boundary conditions are

$$z = 0, 2c/a; w_i = 0 \quad (17)$$

$$r = 0; u_i, v_i \text{ and } w_i \text{ are bounded.}$$

At the interface, they are

$$\begin{aligned} u_{if} &= u_{of} \\ p_{if} &= p_{of} \end{aligned} \quad (18)$$

We once again assume perturbations due to radial displacement,  $\eta$ , of the interface from its equilibrium position corresponding to  $r = e$ .

$$\begin{aligned} u_i &= u'_i \\ v_i &= v'_i \\ w_i &= w'_i \\ p_i &= p'_i + p_1 \end{aligned} \quad (19)$$

Substituting in Equation 3,

$$\begin{aligned} \frac{\partial u'_i}{\partial t} &= - \frac{\partial p'_i}{\partial r} \\ \frac{\partial v'_i}{\partial t} &= - \frac{1}{r} \frac{\partial p'_i}{\partial \theta} \\ \frac{\partial w'_i}{\partial t} &= - \frac{\partial p'_i}{\partial z} \\ \frac{u'_i}{r} + \frac{\partial u'_i}{\partial r} + \frac{1}{r} \frac{\partial v'_i}{\partial \theta} + \frac{\partial w'_i}{\partial z} &= 0 \end{aligned} \quad (20)$$

The boundary conditions become

$$z = 0, \frac{2c}{a}, w'_i = 0 \quad (21)$$

$r = 0$ ;  $u'_i$ ,  $v'_i$  and  $w'_i$  are bounded.

At the interface

$$\begin{aligned} u'_i &= \frac{\partial \eta}{\partial t} \\ p'_{if} &= p_{of} - p_1 \end{aligned} \quad (22)$$

### NORMAL MODES

In accordance with the usual procedure of treating characteristic value problems, the perturbations are analyzed into normal modes. In view of the boundary conditions, it is natural to suppose that the perturbations are given by quantities which have a  $(r, \theta, z, t)$  dependence given by

$$\begin{aligned} u'_o &= U_o(r) \cos [h_o \pi az/2c] e^{i(K_o t + m_o \theta)} \\ v'_o &= B_o(r) \cos [h_o \pi az/2c] e^{i(K_o t + m_o \theta)} \\ w'_o &= W_o(r) \sin [h_o \pi az/2c] e^{i(K_o t + m_o \theta)} \\ p'_o &= R_o(r) \cos [h_o \pi az/2c] e^{i(K_o t + m_o \theta)} \end{aligned} \quad (23)$$

Where  $K_o$  is a constant (which can be complex),  $m_o$  is an integer (which can be positive, zero, or negative) and  $h_o$  is the wave number in the  $z$ -direction (which can be 1, 2, 3, etc.).  $U_o$ ,  $B_o$ ,  $W_o$  and  $R_o$  are functions of  $r$  only.

In an analogous manner, for the interior zone, the perturbation quantities may be assumed to be of the form

$$\begin{aligned}
 u'_i &= U_i(r) \cos(h_i \pi az/2c) e^{i(K_i t + m_i \theta)} \\
 v'_i &= B_i(r) \cos(h_i \pi az/2c) e^{i(K_i t + m_i \theta)} \\
 w'_i &= W_i(r) \sin(h_i \pi az/2c) e^{i(K_i t + m_i \theta)} \\
 p'_i &= R_i(r) \cos(h_i \pi az/2c) e^{i(K_i t + m_i \theta)}
 \end{aligned} \tag{24}$$

where  $K_i$ ,  $m_i$ , and  $h_i$  are defined as before.  $U_i$ ,  $B_i$ ,  $W_i$ , and  $R_i$  are once again assumed to be functions of the radius  $r$  only.

Substituting Equation 23 into Equation 12, we obtain

$$\begin{aligned}
 i(K_o + m_o \frac{V}{r})U_o - \frac{2V}{r}B_o &= -\frac{dR_o}{dr} \\
 \left(\frac{dV}{dr} + \frac{V}{r}\right)U_o + i(K_o + m_o \frac{V}{r})B_o &= -i m_o \frac{R_o}{r} \\
 (K_o + m_o \frac{V}{r})W_o &= -\frac{h_o \pi a}{2c} i R_o \\
 \left(\frac{dU_o}{dr} + \frac{U_o}{r}\right) + i m_o \frac{B_o}{r} + \frac{h_o \pi a}{2c} W_o &= 0
 \end{aligned} \tag{25}$$

In a similar manner, substituting Equation 24 into 20, we obtain

$$\begin{aligned}
 U_i &= \frac{i}{K_i} \frac{dR_i}{dr} \\
 B_i &= -\frac{1}{r} \frac{m_i}{K_i} R_i \\
 W_i &= -\frac{i}{K_i} \frac{h_i \pi a}{2c} R_i \\
 \frac{U_i}{r} + \frac{dU_i}{dr} + \frac{im_i}{r} B_i + \frac{h_i \pi a}{2c} W_i &= 0
 \end{aligned} \tag{26}$$

Since

$$u'_{of} = u'_{if}$$

we have

$$\begin{aligned} U_i(e + n) \cos(h_i \pi az/2c) e^{i(K_i t + m_i \theta)} \\ = U_o(e + n) \cos(h_o \pi az/2c) e^{i(K_o t + M_o)} \end{aligned} \quad (27)$$

Inasmuch as  $\theta$ ,  $z$  and  $t$  are independent variables, we require that  $K_i$ ,  $m_i$  and  $h_i$  be equal to  $K_o$ ,  $m_o$  and  $h_o$  respectively. Thus, the subscripts for  $K$ ,  $m$  and  $h$  will be omitted in what follows. Equation 25 can be simplified to the following

$$\begin{aligned} \sigma \left[ \frac{dU_o}{dr} + \frac{U_o}{r} \right] - \frac{m}{r^2} U_o \frac{d}{dr} (\Omega r^2) &= \left[ \frac{m^2}{r^2} + \frac{\pi^2 h^2 a^2}{4c^2} \right] i R_o \\ \sigma^2 U_o - \frac{2\Omega}{r} U_o \frac{d}{dr} (\Omega r^2) &= \left[ 2\Omega m \frac{R_o}{r} + \sigma \frac{dR_o}{dr} \right] i \end{aligned} \quad (28)$$

where

$$\Omega = V/r \quad (29)$$

and

$$\sigma = K + m\Omega \quad (30)$$

In a similar manner, Equation 26 becomes

$$\begin{aligned} \frac{U_i}{r} + \frac{dU_i}{dr} &= \frac{i}{K} \left[ \frac{m^2}{r^2} + \frac{\pi^2 h^2 a^2}{4c^2} \right] R_i \\ U_i &= \frac{i}{K} \frac{dR_i}{dr} \end{aligned} \quad (31)$$

The boundary conditions are

$$U_o(1) = 0 \quad (32)$$

$U_i(0)$  and  $R_i(0)$  are bounded

The interface conditions are obtained as follows: Since  $P_{of}$  is equal to  $P_{if}'$ , from Equations 15 and 22,

$$P_{if}' = P_{of}' \quad (33)$$

Substituting for  $P'_{of}$  and  $P'_{if}$  from Equations 23 and 24 respectively,

$$\begin{aligned} R_i(e) \cos(h\pi az/2c) e^{i(Kt + m\theta)} \\ = R_o(e) \cos(h\pi az/2c) e^{i(Kt + m\theta)} \end{aligned} \quad (34)$$

where we consider  $\eta \ll e$ .

$$\left[ K \left\{ \frac{U_i}{r} + \frac{dU_i}{dr} \right\} - \sigma \left\{ \frac{dU_o}{dr} + \frac{U_o}{r} \right\} - \frac{m}{r^2} U_o \frac{d}{dr} (\Omega r^2) \right]_{r=e} \quad (35)$$

$$U_o(e) = U_i(e) \quad (36)$$

Thus, the characteristic value problem consists of Equations 28 and 31 with boundary conditions 32, 35, and 36. Since we are interested only in those motions which produce a couple on the casing, interest centers around values of  $h = (2j + 1)$ , where  $j$  is an integer.

#### Axisymmetric Cases \*

Setting  $m = 0$  reduces Equations 28 and 31 to the following

$$\begin{aligned} \frac{d^2 U_o}{dr^2} + \frac{1}{r} \frac{dU_o}{dr} - \left[ \frac{1}{r^2} + N^2 - \frac{N^2}{K^2} \frac{2\Omega}{r} \frac{d}{dr} (\Omega r^2) \right] U_o &= 0 \\ \frac{d^2 U_i}{dr^2} + \frac{1}{r} \frac{dU_i}{dr} - \left[ \frac{1}{r^2} + N^2 \right] U_i &= 0 \end{aligned} \quad (37)$$

$$\text{where } N^2 \equiv \pi^2 h^2 a^2 / 4c^2 \quad (38)$$

The boundary conditions become

$$\begin{aligned} U_o(1) &= 0 \\ U_o(e) &= U_i(e); \quad \frac{dU_o}{dr}(e) = \frac{dU_i}{dr}(e) \\ U_i(0) &= 0 \end{aligned} \quad (39)$$

\* The axisymmetric case is of academic interest only.

Case 1, The Narrow Gap Approximation for  $m = 0$ 

In specifying  $\Omega$ , interest is in those velocity distributions which are realizable in a viscous fluid. Equations 9 and 29 determine the case of primary interest as

$$\Omega = (1 - e^2/r^2)/(1 - e^2) \quad (40)$$

However, as  $e \rightarrow 1$ , an important simplification in the characteristic value problem given by Equations 37, 39 and 40 is possible provided that  $(1 - e)$  is small when compared with  $(1 + e)/2$ . In this case, Equation 40 may be expressed as

$$\Omega = s \quad (41)$$

$$s = (r - e)/(1 - e) \quad (42)$$

The first of Equation 37 may be expressed as

$$\left[ \frac{d^2}{dr^2} + \frac{1}{r} \frac{d}{dr} - \frac{1}{r^2} - N^2 \right] U_o = - \frac{N^2}{K^2} \left[ 2\Omega r \frac{d\Omega}{dr} + 4\Omega^2 \right] U_o \quad (43)$$

Using the transformation defined by Equation 42, consistent with the "narrow gap" approximation of small  $(1 - e)$ , equation 43 becomes

$$\left[ \frac{d^2}{ds^2} - a_1^2 \right] U_o = - \frac{a_1^2}{K^2} \cdot \frac{2e}{1-e} s U_o \quad (44)$$

where

$$a_1^2 \equiv (1 - e)^2 N^2 \quad (45)$$

By means of the following transformation,

$$x = a_1^{2/3} b^{2/3} \left[ \frac{1}{b^2} - s \right] \quad (46)$$

where

$$b^2 \equiv 2e/K^2(1 - e) \quad (47)$$

Equation 44 may be expressed as

$$\frac{d^2 U_o}{dx^2} - x U_o = 0 \quad (48)$$

The general solution of Equation 48 may be given in terms of either Bessel Functions of order 1/3 or somewhat more conveniently in terms of Airy's functions  $AI(x)$  and  $BI(x)$ ; thus,

$$U_o = A_o AI(x) + B_o BI(x) \quad (49)$$

where  $A_o$  and  $B_o$  are the integration constants.

In order to determine  $U_i$ , the general solution of the second of the differential Equation 37 may be obtained from

$$\frac{d^2 U_i}{dR^2} + \frac{1}{R} \frac{dU_i}{dR} - \left[ \frac{1}{R^2} + 1 \right] U_i = 0 \quad (50)$$

$$\text{where } R^2 = N^2 r^2 \quad (51)$$

The general solution of Equation 50 may be written as

$$U_i = A_i I_1(R) + B_i K_1(R) \quad (52)$$

The boundary conditions in terms of the new independent variable  $x$  defined by Equation 46 are

$$U_o(x_1) = 0 \quad (53a)$$

$$U_o(x_2) = U_i(Ne) \quad (53b)$$

$$-\frac{dU_o}{dx} = \frac{N(1-e)}{a_1^{2/3} b^{2/3}} \frac{dU_i}{dR} \text{ at } x = x_2; R = Ne \quad (53c)$$

$$U_i(0) = 0 \quad (53d)$$



where

$$\begin{aligned}x_1 &\equiv a_1^{2/3} b^{2/3} \left[ \frac{1}{b^2} - s \right] \\x_2 &\equiv a_1^{2/3} b^{2/3} / b^2\end{aligned}\quad (54)$$

From boundary conditions 53d, we have

$$A_i I_1(0) + B_i K_1(0) = 0$$

Therefore,

$$B_i = 0 \quad (55)$$

Substituting the general solutions given by Equations 49 and 52 and the condition given by Equation 55 into the boundary conditions 53a, 53b and 53c, we obtain

$$A_o A I(x_1) + B_o B I(x_1) = 0 \quad (56)$$

$$A_o A I(x_2) + B_o B I(x_2) = A_i I_1(Ne) \quad (57)$$

$$\begin{aligned}& - \frac{a_1^{2/3} b^{2/3}}{a_1} \left[ A_o A I'(x_2) + B_o B I'(x_2) \right] \\&= A_i \left[ I_o(Ne) - \frac{I_1(Ne)}{Ne} \right]\end{aligned}\quad (58)$$

where the primes refer to differentiation with respect to  $x$ .  $A_o$ ,  $B_o$  and  $A_i$  may be eliminated from Equations 56, 57 and 58 to give the characteristic equation as

$$x_2^{-1/2} \left[ \frac{A I'(x_2) B I(x_1) - B I'(x_2) A I(x_1)}{A I(x_1) B I(x_2) - A I(x_2) B I(x_1)} \right] = \frac{I_o(Ne)}{I_1(Ne)} - \frac{1}{Ne} \quad (59)$$

From the definition of  $x_1$ , and  $x_2$  in Equation 54, we obtain

$$x_1 = x_2 - N(1 - e)/x_2^{1/2} \quad (60)$$

For given  $N$  and  $e$ , the values of  $x_1$  and  $x_2$  which satisfy Equations 59 and 60 simultaneously are the desired values. However, not all pairs of  $x_1$  and  $x_2$  lead to admissible values of  $K^2$ . From Equation 61 we observe that since

$$K^2 = \frac{2e}{1-e} \cdot \frac{x_2}{x_2 - x_1} = \frac{2e}{N(1-e)^2} x_2^{3/2} \quad (61)$$

and since  $K^2$  is a real number, first of all  $x_2$  must be a positive number. Secondly,  $(x_2 - x_1)$  must also be positive. Thus, of all the values which satisfy Equations 59 and 60, we select only those pairs of  $x_1$  and  $x_2$  values which satisfy the conditions that  $x_2 > 0$  and that  $x_2 > x_1$ .

#### Case ii, The Formal Solution for $m = 0$

In case we wish to consider the complete range of  $e$  in  $1 \geq e \geq 0$ , we must consider Equation 37 without making any approximation. We then have

$$\frac{d^2 U_o}{dr^2} + \frac{1}{r} \frac{dU_o}{dr} - \left[ \frac{v^2}{r^2} + \alpha^2 \right] U_o = 0 \quad (62)$$

where

$$v^2 \equiv 1 + \frac{4N^2 e^2}{K^2 (1 - e^2)^2} \quad (63)$$

$$\alpha^2 \equiv N^2 - \frac{4N^2}{K^2 (1 - e^2)^2} \quad (64)$$

Equation 62 would need to be considered with the second of Equation 37 and boundary conditions 39. The general solution of Equation 62 is

$$U_o = l_v(\alpha r) \quad (65)$$

where  $l$  is a general cylinder function of order  $v$ . The boundary conditions require  $l_v$  to vanish at  $r = 1$ . The required solution may be expressed as

$$U_o = M [J_{-v}(\alpha) J_v(\alpha r) - J_v(\alpha) J_{-v}(\alpha r)] \quad (66)$$

where  $M$  is a constant.

Defined in this manner,  $U_o$  clearly vanishes at  $r = 1$ . We have further at  $r = e$ .

$$M[J_{-v}(\alpha) J_v(\alpha e) - J_v(\alpha) J_{-v}(\alpha e)] = A_i I_1(Ne) \quad (67)$$

Moreover,

$$\left. \frac{dU_o}{dr} \right|_{r=e} = A_i N \left[ I_o(Ne) - \frac{I_1(Ne)}{Ne} \right] \quad (68)$$

The integration constants  $M$  and  $A_i$ ; may be eliminated from Equations 67 and 68 to yield the characteristic equation. For arbitrarily assigned  $v$ , one may compute  $K^2$  from the characteristic equation and Equations 63 and 64 for given  $N$  and  $e$ .

### THE GENERAL CASE

Consistent with boundedness of the solutions of  $U_i$  and  $R_i$  at the origin Equation 31 has the following solution

$$R_i = AI_m(Nr) \quad (69)$$

$$U_i = \frac{i}{K} R'_i = A \left[ \frac{m}{r} I_m + NI_{m+1} \right] \quad (70)$$

where  $A$  is any arbitrary constant. Equation 28 may be rewritten as

$$U'_o = \frac{1}{\sigma} \left[ \left( \frac{m^2}{r^2} + N^2 \right) R_o + \frac{2m}{1-e^2} \cdot \frac{1}{r} U_o \right] - \frac{U_o}{r} \quad (71)$$

$$R'_o = - \frac{2}{\sigma} \left[ \frac{2}{1-e^2} U_o + \frac{m}{r} R_o \right] + \sigma U_o \quad (72)$$

The boundary conditions become

$$U_0(e) = \frac{A}{K} \left[ \frac{m}{e} I_m(Ne) + N I_{m+1}(Ne) \right] \quad (73)$$

$$R_0(e) = A I_m(Ne) \quad (74)$$

$$U_0(1) = 0 \quad (75)$$

where  $R_0$  is redefined as  $-iR_0$  of equations prior to Equation 69. The eigenvalue problem now is the following: find those values of K for which the Equations 71 and 72, and the boundary conditions 73, 74, and 75 are satisfied.

Since Equations 71 and 72 are homogeneous, any constant times any solution that satisfies the boundary conditions will also be a solution. Thus, in the eigenvalue problem, we can arbitrarily assign any value to the constant A. For convenience, we set it equal to unity.

### NUMERICAL SOLUTION

Since Equations 71 and 72 are too cumbersome for closed solutions, we employ numerical integration methods. The numerical procedure for solving the eigenvalue problem is as follows: for an assumed value of K, we can compute the values of  $U_0(e)$  and  $R_0(e)$  for a given set of values of m, e, and N. Using these as the initial values, we can integrate Equations 71 and 72 by numerical methods to yield  $U_0(1)$ . If this value is zero, then our choice of K is indeed the right one. If not, we change our assumed value of k by a selected increment and recompute. In practice, since it is uneconomical, computer time-wise, to calculate a precise zero value for  $U_0(1)$ , we arbitrarily say that if the absolute value of  $U_0(1)$  is less than a prescribed positive small number, then our choice of the k value is the right one.

For negative values of m, Equations 71 and 72 have a singularity depending on whether or not the quantity  $1 + (1 - e^{-2}) K/m$  is real or

complex. If the quantity is real, the singularity is at a radius given by

$$r_{\text{singular}} = e / \sqrt{1 + (1 - e^2) K/m} \quad (76)$$

At this radius, the derivatives of  $U_o$  and  $R_o$  cannot be computed from Equations 71 and 72. However, from physical considerations, we require that  $U_o$  and  $R_o$  be continuous at this point.

When a singularity exists, we employ the following procedure: We divide the two zones,  $e \leq r \leq r_{\text{sing}}$  and  $r_{\text{sing}} \leq r \leq 1$  into  $N_1$  and  $N_2$  intervals, where  $N_1$  and  $N_2$  are selected to be such that the increments in radius for an interval is approximately equal in both zones. Starting from  $r = e$ , we integrate the Equations 71 and 72 till we reach the last interval of  $e \leq r \leq r_{\text{sing}}$ . We subdivide this into four smaller intervals and integrate the equations to obtain the values of  $U_o$ ,  $R_o$  and their derivatives at the three intermediate points shown in Figure 1. Using these values, by linear extrapolation, we estimate the values of  $U_o$ ,  $R_o$  and their derivatives at the singularity. Since we cannot use Equations 71 and 72 at the singularity, using the estimated derivatives, we integrate numerically to the one-fourth point of the next interval. From here on, we revert to the conventional integration procedure using Equations 71 and 72.

The numerical integration procedure used in the computer program for calculating the eigenvalues is the familiar Runge-Kutta-Gill method.

#### DESCRIPTION OF THE COMPUTER PROGRAM

For computing an eigenvalue, the input data to the program is supplied by two cards. The input map for the program, describing the breakdown of the entries in the cards are shown in Table 1. The third I-field on the first card of each set was originally intended to supply the value of  $n$ . Since, this was subsequently generalized to be a

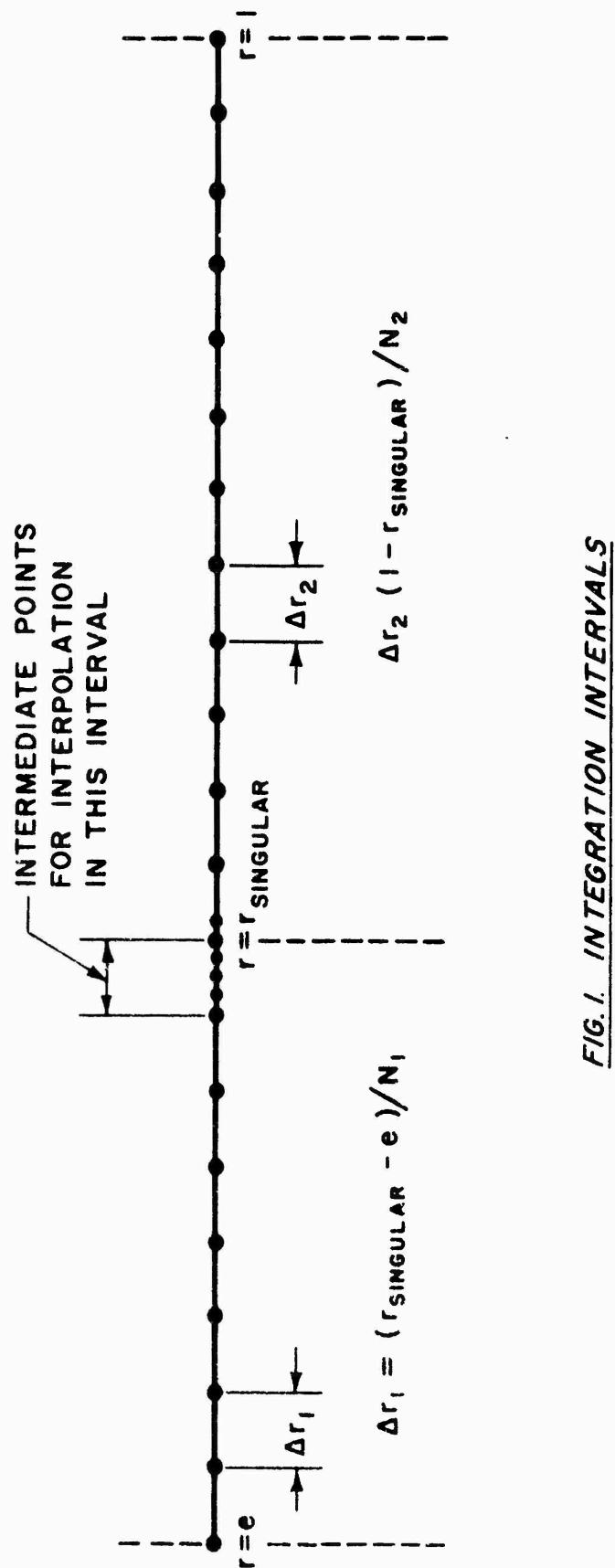


FIG. I. INTEGRATION INTERVALS

Table 1  
INPUT MAP

	<u>Variable</u>	<u>Definition</u>	<u>Format</u>
Card No. 1	M	Run Number	I6
	MM	-m	I6
	NN	Enter zeros	I6
	NR	Number of integration intervals $\leq$ 500	I6
	IGEN	Number of Eigenvalues required	I6
	NSTOP	If last set of data being read, should be entered non-zero. Otherwise, enter zero.	I6
	LC	10 Ne $\leq$ 50	I6
Card No. 2	PLAO	Initial trial value of Eigenvalue	E12.0
	DPLA	Increment of Eigenvalue	E12.0
	TRUNC	Permissible truncation difference	E12.0
	CRAD	e	F12.6

factor in N (CAPN in the program), it may be filled in with a zero or any other number. This number will appear on the printout as the value of n (N in the program). Since it has no other effect on the program, it may be ignored for all other purposes.

The computer program, written in Fortran IV, is given in the Appendix: the first subroutine, NING(NR1) provides the Runge-Kutta-Gill integration method. It also calls a subroutine, DER(Y, FM, C<sub>0</sub>N, CRAD, FK, AK) which provides the derivatives of U<sub>0</sub> and R<sub>0</sub> as given by Equations 71 and 72. Subroutine NUESTI, called by the main program, provides a method for incrementing the eigenvalue on the basis of current derivation in U<sub>0</sub>(1) from the specified truncation error. Subroutine RKGC<sub>0</sub>N (AIN, BIN, CIN) supplies the RKG constants to the main program. Subroutine BESC<sub>0</sub>N(AI, BI), called by the main program supplies the values of the modified Bessel functions. If the program is to be run for any values of m other than -1, this subroutine will need to be changed to supply the modified Bessel functions of the appropriate order.

## RESULTS

Table 2 shows the effect of various step sizes on the computed eigenvalues. Truncation is of  $10^{-6}$ , initial eigenvalue is 0.01, eigenvalue increment is .05.

Table 2  
EFFECT OF STEP SIZE ON COMPUTED EIGENVALUES

Total No. of Steps	Lowest eigenvalue K for Ne = .2 b/a = e = 0.45 m = -1	Lowest eigenvalue K for Ne = .2 b/a = e = .1 m = -1
100	.76079984	.35125716
200	.75407226	.35046438
300	.75447285	.35032046
400	.75445469	.35025621
500	.75479767	.35021697

The results show that if 200 or more total number of steps are employed, the variation is in the fourth significant digit.

Table 3 lists the results for various Ne and e values computed with 200 integration steps, an initial eigenvalue of 0.01, and an eigenvalue increment of 0.05. With this sweep procedure, the listed eigenvalues were found to be the lowest. It is possible that if the eigenvalue increment is lowered from 0.05 to some smaller value, other, ever lower, eigenvalues may be discovered; but this is considered to be extremely unlikely.

In Stewartson's\* notation, we have

$$\frac{c}{a(2j + 1)} = \frac{\pi}{2N}$$

$$b/a = e$$

$$\tau = K$$

\*Stewartson, K., *On The Stability of a Spinning Top Containing Liquid*, J. Fluid Mech., 5, 1959, pp. 577-592.

Table 3  
LOWEST EIGENVALUES

	$\frac{\pi}{2} \frac{a(2j+1)}{c}$	$b/a$	$K = \tau$	$\frac{\pi}{2}$	$\frac{10XNe}{c}$	$\frac{\pi}{2} \frac{a(2j+1)}{c}$	$b/a$	$K = \tau$	$n$
1	2	.05	.35294417	2	3	6	.05	0.10161652	4
	1	.1	.36637878	1		3	.1	0.079672609	2
	.5	.2	.68914656	1		1.5	0.2	0.077654683	1
	.333	.3	.78440604	1		1	0.3	0.38693088	1
	.25	.4	.95493352	1		.75	0.4	0.53024257	1
	.2	.5	.98246679	1		.6	0.5	0.65401489	1
	.166	.6	.99240964	1		.5	0.6	0.93675748	1
	.142857	.7	.99677785	1		.42871	0.7	0.96985291	1
	.125	.8	.99793433	1		.375	0.8	0.98589012	1
	.1111	.9	.99749419	1		.333	0.9	0.99401352	1
2	4	.05	.17450405	3	4	8	.05	0.062517198	5
	2	.1	.35046438	2		4	.1	0.1748773	3
	1	.2	.37567224	1		2	.2	0.33798441	2
	.666	.3	.57999146	1		1.33	.3	0.19826175	1
	.5	.4	.68582012	1		1	0.4	0.39889403	1
	.4	.5	.93226799	1		.8	0.5	0.5155985	1
	.333	.6	.97111366	1		0.6667	0.6	0.69132881	1
	.285714	.7	.98598000	1		0.571429	0.7	0.94881139	1
	.25	.8	.99387097	1		.5	0.8	0.97572948	1
	.222	.9	.99698953	1		0.444	0.9	0.98943233	1

$n$  is the number of radial nodes in the range  $0 < r < 1$



Table 3 (Concl.)

$\frac{\pi}{2} \frac{a(2j+1)}{c}$	$b/a$	$k = \tau$	$\frac{\pi}{2} \frac{a(2j+1)}{c}$	$b/a$	$k = \tau$
5	10	.05	0.03821307	6	7
	5	.1	0.021487953	3	7
2.5	0.2	0.21049394	2	3.5	.1
1.666	0.3	0.028242248	1	2.333	.2
1.25	0.4	0.27032753	1	1.75	.3
1	0.5	0.41369377	1	1.4	.4
.833	0.6	0.54600539	1	1.166	.5
.714268	0.7	0.9210945	1	1	.6
.625	0.8	0.96316761	1	.875	.7
.555	0.9	0.98512214	1	.777	.8
					.97391060
6	12	.05	0.021654819	7	8
	6	.1	0.1034529	4	8
3	.2	0.09 <sup>4</sup> 563088	2	4	.1
2	.3	0.32489160	2	2.666	.2
1.5	.4	0.14853735	1	2	.3
1.2	.5	0.32126868	1	1.6	.4
1	.6	0.44177624	1	1.333	.5
.857143	.7	0.69269504	1	1.142857	.6
.75	.8	0.94963371	1	1	.7
.666	.9	0.97912326	1	.888	.8
					.96724922

n is the number of radial nodes in the range 0 < r < 1

## **APPENDIX**

A

REDDI

04/01/66

MAIN

- EFN SOURCE STATEMENT - IFN(S) -

## MAIN PROGRAM FOR THE DETERMINATION OF EIGENVALUES 31G-B2294-01

```

      COMMON FM,FN,CON,DEL,NR,X(1,501),Z(1,501),FREQU(3),DET(3),AIN(4),
     1BIN(4),CIN(4),CRAD,DPLA,IDEF,ILU,KLUE,UC0,RC0,RADIUS(501)
      DIMENSION AI(50),BI(50)
      NSTOP=0
      CALL RKGCON(AIN,BIN,CIN)
      CALL BESCON(AI,BI)
20    IF(NSTOP.NE.0)STOP
      READ(5,10)M,MM,NN,NR,IGEN,NSTOP,LC,PLAO,DPLA,TRUNC,CRAD
10    FORMAT(7I6/3E12.0,F12.6)
      MM=-MM
      FM=MM
      FN=NN
      FNR=NR
      FREQU(1)=PLAO
      BNC=LC
      FNC=BNC/10.
      CAPN=FNC/CRAD
      CON=CAPN**2
      AIM1=AI(LC)*EXP(FNC)
      AIM2=BI(LC)*EXP(FNC)
      IDET=1
      KKK=1
      WRITE(6,85)
85    FORMAT(1H1 35HCURRENT VALUES OF INTEGRATION DATA //)
      WRITE(6,86)CAPN,LC
86    FORMAT(3X7HCAPN = F10.4,5X5HLC = I2)
      DO 30 IG=1,IGEN
      ICOUNT=1
70    UC0=(FM*AIM1/CRAD+CAPN*AIM2)/FREQU(1)
      RC0=AIM1
      RADIUS(1)=CRAD
      X(1,1)=UC0
      Z(1,1)=RC0
      CALL NING(NR1)
      DET(1)=X(1,NR+1)
      WRITE(6,90)M,IG,FREQU(1),X(1,NR+1)
90    FORMAT(                                // 10X 13HRUN NUM
      1BER = I3, 5X 5HIG = I2, 5X 16HTRIAL EIGENV. = E15.8, 5X 9H U(1) 00000023
      2= E15.8)                                         00000024
      IF(IDEF=2)170,170,180                           00000025
170   DL = DET(1)
      AL = FREQU(1)
      GO TO 190
180   IF(AL=FREQU(1))170,170,190                  00000029
190   ADET = ABS(DET(1))                            00000030
      IF(ADET=TRUNC)80,80,100                         00000031
100   CALL NUESTI                                    00000032
      GO TO (80,110, 20),KLUE                         00000035
110   IDET = IDET+1                                 00000036
      ICOUNT=ICOUNT+1
      IF(ICOUNT=50)70,70, 20
80    WRITE(6,1)
1   FORMAT(1H1 41X 32I EIGENVALUES AND EIGENFUNCTIONS ) 00000040

```

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MAIN

- IFN SOURCE STATEMENT - IFN(S) -

```

      WRITE(6,2)M,MM,NN,IG,FREQU(1),CAPN,CRAD
2 FORMAT(//48X 10HR IN NO. = I3// 7X 6H M = I3, 5X 8H N = I3,
1   5X I2, 2X 13H RIGENVALUE = E15.8,9X6HCAPN = F12.6,5X4HC = F6.2)
      WRITE(6,3)
3 FORMAT(///18X 2H R 4X 14H U MODE SHAPE 4X 14H R MODE SHAPE 18X 2H
5R 4X 14H U MODE SHAPE 3X 15H R MODE SHAPE //)
      KPA=1          00000047
      NRR=NR/2        00000048
      DO 40 I=1,NRR  00000049
      IF (I=51*KPA) 50,60,50
60 KPA = KPA+1    00000051
      WRITE(6,1)
      WRITE(6,2)M,MM,NN,IG,FREQU(1),CAPN,CRAD
      WRITE(6,3)
50 K=NRR+1        00000055
      WRITE(6,130) RADIUS(I),X(1,I),Z(1,I),RADIUS(K),X(1,K),Z(1,K)
130 FORMAT(F20.4, 3X E15.8, 3X E15.8,F20.4, 3X E15.8, 3X E15.8) 00000057
      IF(I-NRR) 40,140,40  00000058
140 K = K+1        00000059
      WRITE(6,150) RADIUS(K),X(1,K),Z(1,K)
150 FORMAT(40X F20.4,3X E15.8, 3X E15.8)
40 CONTINUE        00000062
      FREQU(2)=AL    00000063
      DFT(2) =DL    00000064
      FREQU(1)=AL+DPLA 00000065
30 IDET=2          00000066
      GO TO 20        00000067
      END             00000068

```

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05/02/66

GOODY

- EFN SOURCE STATEMENT - IFN(S) -

```
SUBROUTINE NING(NR1)
COMMON FM,FN,CON,DEL,NR,X(1,501),Z(1,501),FREQU(3),DET(3),AIN(4),
1BIN(4),CIN(4),CRAD,DPLA,DET,ILU,KLUE,UC0,RC0,RADIUS(501)
DIMENSION AK(3),Y(3),Q(3),AM(3),BM(3)
FK=FREQU(1)
Y(1)=CRAD
Y(2)=UC0
Y(3)=RC0
Q(1)=0,
Q(2)=0,
Q(3)=0,
AK(1)=1,
BING=1.+1.-(CRAD**2))*FK/FM
IF (BING)3,3,1
1 RSING=CRAD/SQRT(BING)
WRITE(6,2) RSING
2 FORMAT(//20X 14HR(SINGULAR) = F9.5)
GO TO 4
3 WRITE(6,5)
5 FORMAT(10X38HSINGULARITY OUTSIDE INTEGRATION DOMAIN      )
4 BNR=NR
NR1=BNR*(RSING-CRAD)/(1.-CRAD)
NRM=NR-50
IF(NR1.LT.50)NR1=50
IF(NR1.GT;NRM)NR1=NRM
NR2=NR-NR1
FNR1=NR1
FNR2=NR2
DEL=(RSING-CRAD)/FNR1
DEL1=(1.-RSING)/FNR2
DELT=DEL/4.
NAB=NR1+2
DO 1000 L=1,NAB
IF(L.GE.NR1)DEL=DELTA
DO 100 JJ=1,4
CALL DER(Y,FM,CON,CRAD,FK,AK)
DO 50 I=1,3
AIKN =AIN(JJ)*(AK(I)-BIN(JJ)*Q(I))
Y(I) = Y(I)+DEL*AIKN
50 Q(I) = Q(I)+3.*AIKN -CIN(JJ)*AK(I)
100 CONTINUE
RADIUS(L+1)=Y(1)
X(1,L+1)=Y(2)
1000 Z(1,L+1)=Y(3)
CALL DER(Y,FM,CON,CRAD,FK,AK)
AU=AK(2)
AR=AK(3)
AK(2)=2.*BU-AU
AK(3)=2.*BR-AR
Y(1)=RSING
Y(2)=2.*X(1,NAB+1)-X(1,NAB)
```

RENDI

03/02/66

GOODY - EFN SOURCE STATEMENT - IFN(S) -

```
Y(3)=2.*Z(1,NAB+1)-Z(1,NAB)
RADIUS(NR1+1)=Y(1)
X(1, NR1+1)=Y(2)
Z(1, NR1+1)=Y(3)
DEL=DEL1/4,
DO 80 L=1,4
DO 80 JJ=1,4
IF(L.EQ.1)GO TO 81
CALL DER(Y,FM,CON,CRAD,FK,AK)
81 DO 82 I=1,3
AIKN =AIN(JJ)*(AK(I)-BIN(JJ)*Q(I))
Y(I) = Y(I)+DEL*AIKN
82 Q(I) = Q(I)+3.*AIKN -CIN(JJ)*AK(I)
80 CONTINUE
RADIUS(NR1+2)=Y(1)
X(1, NR1+2)=Y(2)
Z(1, NR1+2)=Y(3)
DEL=DEL1
DO 900 L=NAB,NR
IF(L.EQ.NR)DEL=1.-Y(1)
DO 90 JJ=1,4
CALL DER(Y,FM,CON,CRAD,FK,AK)
DO 60 I=1,3
AIKN =AIN(JJ)*(AK(I)-BIN(JJ)*Q(I))
Y(I) = Y(I)+DEL*AIKN
60 Q(I) = Q(I)+3.*AIKN -CIN(JJ)*AK(I)
90 CONTINUE
RADIUS(L+1)=Y(1)
X(1,L+1)=Y(2)
900 Z(1,L+1)=Y(3)
RETURN
END
```

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DERIVE - EFN SOURCE STATEMENT - IFN(S) -

```
SUBROUTINE DER(Y,FM,CON,CRAD,FK,AK)
DIMENSION Y(3),AK(3)
R2=1./(Y(1)*Y(1))
ALPHA=FM*FM*R2+CON
BETA=2./(-CRAD*CRAD)
OMEGA=.5*(1.-CRAD*CRAD*R2)*BETA
SIGMA=FK+FM*OMEGA
AK(2)=(ALPHA*Y(3)+FM*BETA*Y(2)/Y(1))/SIGMA-Y(2)/Y(1)
AK(3)=-2.*OMEGA*(BETA*Y(2)+FM*Y(3)/Y(1))/SIGMA+SIGMA*Y(2)
RETURN
END
```

REFDT

POLAR

- EFN SOURCE STATEMENT - IFN(S) -

02/22/66

SUBROUTINE NUESTI  
 COMMON FM, FN, CON, DEL, NR, X(1,501), Z(1,501), FREQU(3), DET(3), AIN(4),  
 TBIN(4), CTN(4), CRAD, DPLA, IDFT, ILU, KLUF, UCO, RCO, RADIUS(501)  
 DOUBLE PRECISION F321, F213, FD, AF1, F23, F31, F12, D1F1, D2F2, D3F3, AF2,  
 FE132, AF3, QD, CC, SR1, FF1, FF2  
 IF(IDFT-2) 10, 20, 30 00000123  
 10 FRFQU(2)=FRFQU(1) 00000124  
 FRFQU(1)=FRFQU(2)+DPLA  
 DET(2)=DET(1) 00000126  
 NCLUF= 3 00000127  
 GO TO 40 00000128  
 20 DDFT = DET(1)\* DET(2) 00000129  
 NCLUF = 3 00000130  
 IF(DDFT) 50, 50, 60 00000131  
 60 ILU = 2 00000132  
 GO TO 70 00000133  
 50 ILU = 1 00000134  
 70 FRFQU(3)=FRFQU(2) 00000135  
 DET(3) = DET(2);  
 FRFQU(2) = FRFQU(1)  
 DET(2) = DET(1) 00000136  
 95 GO TO (90,100), ILU 00000137  
 90 FRFQU(1)=.5\*((FRFQU(3)+FRFQU(2))-((FRFQU(3)-FRFQU(2))/(DET(3)-  
 X DET(2)))\*(DET(3)+DET(2))) 00000138  
 GO TO 40 00000139  
 100 FRFQU(1)=FRFQU(2)+DPLA  
 GO TO 40 00000140  
 1 GO TO (110,120), ILU 00000141  
 110 DDFT = DET(1)\*DET(2) 00000142  
 IF(DDFT) 130, 130, 140 00000143  
 130 NCLUF = 1 00000144  
 GO TO 150 00000145  
 140 DDFT=DET(1)\*DET(3) 00000146  
 NCLUF = 2 00000147  
 IF(DDFT) 150, 150, 70 00000148  
 150 IF(FRFQU(3)) 165, 165, 300 00000149  
 165 GO TO (175,185), NCLUF 00000150  
 175 FRFQU(3)=FRFQU(1) 00000151  
 DET(3)=DET(1) 00000152  
 GO TO 95 00000153  
 185 FRFQU(2)=FRFQU(1) 00000154  
 DET(2)=DET(1) 00000155  
 GO TO 95 00000156  
 300 DF12=ABS ((FRFQU(1)-FRFQU(2))/FRFQU(1)) 00000157  
 DF13=ABS ((FRFQU(1)-FRFQU(3))/FRFQU(1)) 00000158  
 DF23=ABS ((FRFQU(2)-FRFQU(3))/FRFQU(2)) 00000159  
 IF(DF12-.1F-04) 312, 312, 313 00000160  
 313 IF(DF13-.1F-04) 312, 312, 314 00000161  
 314 IF(DF23-.1F-04) 312, 312, 311 00000162  
 312 GO TO (165,501), ILU 00000163  
 311 D1 = DET(1) 00000164  
 D2 = DET(2) 00000165  
 D3 = DET(3) 00000166  
 F1 = FRFQU(1) 00000167  
 F2 = FRFQU(2) 00000168  
 F3 = FRFQU(3) 00000169  
 F4 = FRFQU(4) 00000170  
 F5 = FRFQU(5) 00000171

RF001

02/22/66

PO1 AR - FFN SOURCE STATEMENT - IFN(S) -

F3 = FRFQU(3)	00000172
F321 = F3-F21/F1	00000173
F132 = (F1-F3)/F2	00000174
F213 = (F2-F1)/F3	00000175
FD = F321 + F132 + F213	00000176
AF1 = (D1*F321 + D2*F132 + D3*F213)/FD	00000177
F23 = F2/F3	00000178
F31 = F3/F1	00000179
F12 = F1/F2	00000180
D1F1 = D1/F1	00000181
D2F2 = D2/F2	00000182
D3F3 = D3/F3	00000183
AF2 = (D1F1*(F23-1.0/F23)+D2F2*(F31-1.0/F31)+D3F3*(F12-1.0/F12))/00000184	
1FD	00000185
AF3 = (D1F1*(1.0/F2-1.0/F3)+D2F2*(1.0/F3-1.0/F1)+D3F3*(1.0/F1-1.000000186	
1/F21)/FD	00000187
QP = AF1/AF3	00000188
CC = 0.5*AF2/AF3	00000189
CCD=CC**2-QP	00000190
IF(CCD)312.312.397	00000191
397 SR1 = SORT (CCD)	00000192
FF1 = -CC+SR1	00000193
FF2 = -CC -SR1	00000194
GO TO(160,400),ILU	00000195
400 ILU = 1	00000196
FRFOUT(3) = FRFF	00000197
DF(3) = DET(2)	00000198
FREQU(2) = FRFQU(1)	00000199
DFT(2) = DFT(1)	00000200
	00000201
80 IF(FF1-FRFQU(3))500.500.600	00000202
500 FRFOUT(1)=FF2	00000203
GO TO 40	00000204
600 IF(FF1-FRFQU(2))700.700.500	00000205
700 FRFQU(1)=FF1	00000206
GO TO 40	00000207
160 GO TO(170,180),NCLUF	00000208
170 FRFOUT(3)=FRFOUT(1)	00000209
DET(3) =DET(1)	00000210
GO TO 80	00000211
180 FREQU(2)=FRFQU(1)	00000212
DET(2) =DET(1)	00000213
GO TO 80	00000214
120 DDFT = DFT(1)*DET(2)	00000215
IF(DDFT)190.190.200	00000216
190 NCLUF = 2	00000217
IF(FRFQU(3))50.50.300	00000218
200 NCLUF = 3	00000219
GO TO 70	00000220
40 TU TO(210,220,230),NCLUF	00000221
210 DFRA = ABS (FRFQU(1)-FRFQU(3))	00000222
GO TO 240	00000223
220 DFRA = ABS (FRFQU(1)-FREQU(2))	00000224
230 IF(DFRA-.1F-07)250.250.230	00000225
240 KLUF = 1	00000226
GT TO(260,270,280),NCLUF	00000227

RFOOT

POLAR

- FPN SOURCE STATEMENT - IFN(S) -

02/22/66

260	FRFQU(1) = FRFQU(3)	00000228
	GO TO 280	00000229
1	FRFQU(1) = FRFQU(2)	00000230
	GO TO 280	00000231
230	KIUF = 2	00000232
280	CONTINUF	00000233
	RRETURN	00000234
	FND	00000235

REDDI  
RUNGE

- EFN SOURCE STATEMENT - IFN(S) -

01/27/66

```
SUBROUTINE RKGCON(AIN,BIN,CIN)
DIMENSION AIN(4),BIN(4),CIN(4)
AIN(1)=1./2.
SRT=SQRT(AIN(1))
AIN(2)=1.-SRT
AIN(3)=1.+SRT
AIN(4)=1./6.
BIN(1)=2.
BIN(2)=1.
BIN(3)=1.
BIN(4)=2.
CIN(1)=AIN(1)
CIN(2)=AIN(2)
CIN(3)=AIN(3)
CIN(4)=AIN(1)
RETURN
END
```

REDDI

600F4

- EFN SOURCE STATEMENT - IFN(S) -

02/09/66

SUBROUTINE BESCON(AI,BI)

DIMENSION AI(50),BI(50)

AI(1) = 0.0452984

AI(2) = 0.0822831

AI(3) = 0.1123775

AI(4) = 0.1367632

AI(5) = 0.1564208

AI(6) = 0.1721644

AI(7) = 0.1846699

AI(8) = 0.1944987

AI(9) = 0.2021165

AI(10)=0.2079104

AI(11)=0.2122016

AI(12)=0.2152568

AI(13)=0.2172976

AI(14)=0.2185076

AI(15)=0.2190694

AI(16)=0.2190195

AI(17)=0.2185528

AI(18)=0.2177263

AI(19)=0.2166120

AI(20)=0.2152693

AI(21)=0.2137478

AI(22)=0.2120877

AI(23)=0.2103230

AI(24)=0.2084811

AI(25)=0.2065846

AI(26)=0.2046523

AI(27)=0.2026990

AI(28)=0.2007374

AI(29)=0.1987773

AI(30)=0.1968267

AI(31)=0.1948921

AI(32)=0.1929786

AI(33)=0.1910902

AI(34)=0.1892299

AI(35)=0.1873999

AI(36)=0.1856022

AI(37)=0.1838379

AI(38)=0.1821076

AI(39)=0.1804119

AI(40)=0.1787508

AI(41)=0.1771244

AI(42)=0.1755325

AI(43)=0.1739746

AI(44)=0.1724502

AI(45)=0.1709588

AI(46)=0.1694997

AI(47)=0.1680723

AI(48)=0.1666757

AI(49)=0.1653093

AI(50)=0.1639723

BI(1) = .9071069

BI(2) = .8269385

BI(3) = .7575806

ISUM

ACTUALLY  
ISUB1

ISUBM+1

ACTUALLY

REDDI

GOOF4

- EFN SOURCE STATEMENT - IFN(S) -

02/09/66

BI(4) = .6974022  
BI(5) = .6450353  
BI(6) = .5993272  
BI(7) = .5593055  
BI(8) = .5241489  
BI(9) = .4931630  
BI(10)= .4657596  
BI(11)= .4414404  
BI(12)= .4197821  
BI(13)= .4004249  
BI(14)= .3830625  
BI(15)= .3674336  
BI(16)= .3533150  
BI(17)= .3405157  
BI(18)= .3288719  
BI(19)= .3182432  
BI(20)= .3085083  
BI(21)= .2995631  
BI(22)= .2913173  
BI(23)= .2836930  
BI(24)= .2766223  
BI(25)= .2700464  
BI(26)= .2639140  
BI(27)= .2581801  
BI(28)= .2528055  
BI(29)= .2477557  
BI(30)= .2430003  
BI(31)= .2385126  
BI(32)= .2342688  
BI(33)= .2302480  
BI(34)= .2264314  
BI(35)= .2228024  
BI(36)= .2193462  
BI(37)= .2160494  
BI(38)= .2129001  
BI(39)= .2098875  
BI(40)= .2070019  
BI(41)= .2042345  
BI(42)= .2015774  
BI(43)= .1990232  
BI(44)= .1965656  
BI(45)= .1941983  
BI(46)= .1919160  
BI(47)= .1897134  
BI(48)= .1875862  
BI(49)= .1855300  
BI(50)= .1835408

RRETURN

END

ISUBZERO

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(Security classification of title, body of abstract and indexing annotation must be entered when the overall report is classified)

1. ORIGINATING ACTIVITY (Corporate author) The Franklin Institute Research Laboratories Benjamin Franklin Parkway Philadelphia, Pennsylvania		2a. REPORT SECURITY CLASSIFICATION Unclassified 2b. GROUP
3. REPORT TITLE ON THE EIGENVALUES OF COUETTE FLOW IN A FULLY-FILLED CYLINDRICAL CONTAINER		
4. DESCRIPTIVE NOTES (Type of report and inclusive dates)		
5. AUTHOR(S) (Last name, first name, initial) Reddi, M. M.		
6. REPORT DATE January 1967	7a. TOTAL NO. OF PAGES	7b. NO. OF REFS
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11. SUPPLEMENTARY NOTES	12. SPONSORING MILITARY ACTIVITY U.S. Army Ballistic Research Laboratories Aberdeen Proving Ground, Maryland	
13. ABSTRACT For a stationary flow in a cylindrical container of the Couette type in an outer radial zone, and of zero velocity in an inner radial zone, the normal mode equations are derived. For negative wave numbers in the $\theta$ -direction, these equations are found to have a singularity.  The eigenvalues are calculated by initial value methods employing the Runge-Kutta-Gill integration procedure. Values of the dependent function and their derivatives at the singularity are calculated by linear extrapolation coupled with continuity requirements.  Tables of eigenvalues for various slenderness ratios of the cylinder and various radial nodes are given for $\theta$ -wave numbers of -1.		

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Eigenvalues, Viscous Fluid Couette Flow Rotating Cylinders						
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